

plane with a uniform angular velocity  $\omega$ , which is set as close to  $2\pi/T$  as possible, where  $T$  is the period of the orbit;

(b) a horizon sensor mounted in the cylinder, which measures the angle  $\alpha(t)$  between the line of sight to the Earth's center and a reference line on the rotating cylinder. This reference line is chosen so that  $\alpha(T_0) \cong 0$ ; that is,  $\alpha \cong 0$  at perigee. In the absence of measurement error,  $\alpha(t)$  is a periodic function, well defined by the parameters  $e$ ,  $a$ ,  $T_0$ ,  $\omega$ , and  $\alpha(T_0)$ ; e.g., if the orbit is circular, then  $\alpha(t) = 0$  if  $\omega = 2\pi/T$  exactly. Any deviation of the orbit from a circle will cause the rotating reference line to periodically lead and lag behind the line of sight. The problem here is to improve the estimates of  $e$ ,  $a$ ,  $T_0$ ,  $\omega$ , and  $\alpha(T_0)$ , based on noisy measurements of  $\alpha(t)$ .

The relationship between  $\alpha$ , at a time  $t$ , and the parameters  $a$ ,  $e$ ,  $T_0$ ,  $\omega$ , and  $\alpha(T_0)$  is given *implicitly* as follows:†

$$\alpha = \phi - m; \quad \cos \phi = \frac{\cos E - e}{1 - e \cos E}; \quad M = E - e \sin E;$$

$$M = \frac{2\pi(t - T_0)}{T}; \quad m = \omega(t - T_0) - \alpha_0; \quad \alpha_0 \equiv \alpha(T_0);$$

$$T = \frac{2\pi}{R\sqrt{g}} a^{3/2}.$$

Here the angles  $\phi$ ,  $M$ , and  $E$  are known as the true, mean, and eccentric anomalies respectively,  $g$  = acceleration of gravity at the Earth's surface, and  $R$  = radius of Earth. Note that  $m \equiv M$  if  $\omega = 2\pi/T$  and  $\alpha_0 = 0$ .

By taking *differentials* of the relationships above and eliminating  $d\phi$ ,  $dM$ ,  $dE$ ,  $dm$ , and  $dT$ , the following relation may be obtained:

$$d\alpha = \frac{\partial \alpha}{\partial a} da + \frac{\partial \alpha}{\partial e} de + \frac{\partial \alpha}{\partial T_0} dT_0 + T_0 d\omega + d\alpha_0,$$

where

$$\frac{\partial \alpha}{\partial a} = \frac{3\pi T_0 (1 - e^2) \sin E}{aT \sin \phi (1 - e \cos E)^3}, \quad \frac{\partial \alpha}{\partial e} = \frac{(1 - e^2) \sin^2 E}{\sin \phi (1 - e \cos E)^3}$$

$$\frac{\partial \alpha}{\partial T_0} = \omega - \frac{2\pi}{T} \frac{(1 - e^2) \sin E}{\sin \phi (1 - e \cos E)^3},$$

and these partial derivatives are evaluated with the best current estimates of  $a$ ,  $e$ ,  $T_0$ ,  $\omega$ ,  $\alpha_0$  at the time of the measurement.

The measurement  $z(t)$  is assumed to contain a random error,

†See, for example, J. M. A. Danby, *Fundamental of Celestial Mechanics*, New York: Macmillan, 1962.

ular velocity  $\omega$ , which is set as the  
 s the period of the orbit.  
 nted in the cylinder, which measurement  
 e of sight to the Earth's center and  
 ing cylinder. This reference line  
 at is,  $\alpha \cong 0$  at perigee. In the absence  
 a periodic function, well defined over  
 $\alpha(T_0)$ ; e.g., if the orbit is circular,  $\alpha$   
 Any deviation of the orbit from a  
 erence line to periodically localizing  
 The problem here is to improve the  
 $T_0$ ), based on noisy measurements  
 $\alpha$ , at a time  $t$ , and the parameters  
 plicitly as follows:†

$$= \frac{\cos E - e}{1 - e \cos E}; \quad M = E - e \sin E$$

$$= \omega(t - T_0) - \alpha_0; \quad \alpha_0 \cong \alpha(T_0)$$

are known as the true, mean, and  
 $g$  = acceleration of gravity at the Earth's  
 Earth. Note that  $m \equiv M$  if  $\omega = 2\pi/T_0$

the relationships above and eliminate  
 following relation may be obtained:

$$de + \frac{\partial \alpha}{\partial T_0} dT_0 + T_0 d\omega + d\alpha_0$$

$$\frac{\partial \alpha}{\partial e} = \frac{(1 - e^2) \sin E}{\sin \phi (1 - e \cos E)}$$

are evaluated with the best current  
 e time of the measurement.  
 assumed to contain a random error

Fundamental of Celestial Mechanics

$$z(t) = \alpha(t) + v,$$

where  $E(v) = 0$ ,  $E(v^2) = R$ , and  $R$  is known. Let  $\bar{\alpha}(t)$  be the predicted measurement, using the best current estimates of  $a$ ,  $e$ ,  $T_0$ ,  $\omega$ ,  $\alpha_0$ , at time  $t$ . Then we have

$$z(t) - \bar{\alpha}(t) \cong d\alpha(t) + v$$

$$z(t) - \bar{\alpha}(t) \cong \begin{bmatrix} \frac{\partial \alpha}{\partial a} & \frac{\partial \alpha}{\partial e} & \frac{\partial \alpha}{\partial T_0} & T_0 & 1 \end{bmatrix} \begin{bmatrix} da \\ de \\ dT_0 \\ d\omega \\ d\alpha_0 \end{bmatrix} + v.$$

This linear relation can then be used to estimate  $da$ ,  $de$ ,  $dT_0$ ,  $d\omega$ , and  $d\alpha_0$  from the measurement  $z(t) - \bar{\alpha}(t)$ .

1. In Equation (12.2.1), assume that  $x$  and  $v$  are independent random vectors with gaussian density functions. Show that the joint density function  $p(x, v)$  is proportional to  $\exp(-J)$ , where  $J$  is as defined in Equation (12.2.5). Thus,  $x = \hat{x}$ ,  $v = z - H\hat{x}$  maximize  $p(x, v)$ , giving the name "maximum likelihood estimate."

2. Establish the relations

$$P = M - MH^T(HMH^T + R)^{-1}HM, \tag{a}$$

$$PH^TR^{-1} = MH^T(HMH^T + R)^{-1}. \tag{b}$$

that these relations involve inverting matrices of smaller dimension than  $P$  and  $K$  than Equation (12.2.8) if  $R$  is of smaller dimension than  $P$ ; i.e., if  $p < n$ . Equations (a) and (12.2.8) are known as the "matrix inversion pair" (see Problem 4, Section 1.3).

3. Complete the square in Equation (12.2.5) and show that

$$J = \frac{1}{2} [x - \bar{x} - PH^TR^{-1}(z - H\bar{x})]^T P^{-1} [x - \bar{x} - PH^TR^{-1}(z - H\bar{x})] + \frac{1}{2} (z - H\bar{x})^T R^{-1} (R - HPH^T) R^{-1} (z - H\bar{x}).$$

$J$  is minimized by choosing  $x = \hat{x}$ , where

$$\hat{x} = \bar{x} + PH^TR^{-1}(z - H\bar{x}),$$

which is in agreement with Equation (12.2.7).

4. Given two correlated gaussian random vectors  $x$  and  $z$ , with mean values  $\bar{x}, \bar{z}$  and covariance matrices  $P_{xx}, P_{zz}$ , respectively,

and correlation  $E[(x - \bar{x})(z - \bar{z})^T] = P_{xz}$ , show that the conditional density function  $p(x/z)$  is gaussian, with

$$E(x/z) = \bar{x} + P_{xz}P_{zz}^{-1}(z - \bar{z}) = \hat{x},$$

$$E\{[(x - \hat{x})(x - \hat{x})^T]/z\} = P_{xx} - P_{xz}P_{zz}^{-1}P_{xz}^T.$$

**Problem 5.** In Problem 4, let  $z = Hx + v$ , where  $H$  is a known matrix and  $v$  is independent of  $x$ , with mean value zero and covariance  $R$ . Let  $P_{xx} = M$  and show that

$$P_{zz} = R + HMH^T, \quad P_{xz} = MH^T, \quad \bar{z} = H\bar{x}.$$

Using these relations in Problem 4, verify Equations (12.2.7) and (12.2.8) (you will also need the results of Problem 2). Note that  $K = P_{xz}P_{zz}^{-1}$ , a most reasonable result!

**Problem 6.** In Problem 4, show that the gaussian random vectors  $e = E(x/z) - x$  and  $z - \bar{z}$  are independent; i.e., we have  $E[e(z - \bar{z})^T] = 0$ .

**Problem 7.** Suppose that the number of theoretical relationships among measured variables  $z$  and state variables  $x$  is less than the number of measured variables; for example, we have

$$Az = Hx + Av,$$

where

$$A \text{ is a } (q \times p)\text{-matrix, } q < p, \quad H \text{ is a } (q \times n)\text{-matrix,}$$

$$E(v) = 0, \quad E(vv^T) = R \text{ is a } (p \times p)\text{-matrix.}$$

Show that the estimation procedure of this section applies without replacement by  $Az$  and  $R$  by  $ARA^T$ .

**Problem 8.** Consider the usual problem of least square fit, i.e., determining  $x$  to minimize

$$J = \frac{1}{2}\|z - Hx\|^2.$$

Show that the error of the fit  $e = z - H\hat{x}$  is orthogonal to the fit  $\hat{z} = H\hat{x}$ , in the sense  $e^T\hat{z} = 0$ .

**Problem 9.** In Example 2, suppose that initial estimates of an orbit are  $a = 6,000$  miles,  $\bar{e} = 1/6$ ,  $\bar{T}_0 = 0$ ,  $\bar{\omega} = 2\pi/T$ ,  $\bar{\alpha}_0 = 0$ . Take  $R = 2.56$  miles,  $g = 32.2$  ft sec<sup>-2</sup>, and make an improved estimate of  $a, e, T, \omega$ , and  $\alpha_0$ , using the single measurement

$$z(t) = 16.7 \text{ deg} \quad \text{at} \quad t = 1,357 \text{ sec,}$$

where

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### 12.3 Optimal filtering

Consider a syst

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$E$

Furthermore,  $x_0$

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$\bar{z}^T] = P_{xz}$ , show that the conditional Gaussian, with

$$P_{xz}P_{zz}^{-1}(z - \bar{z}) = \hat{x},$$

$$E\{\hat{x}^T | z\} = P_{xx} - P_{xz}P_{zz}^{-1}P_{zx}.$$

$z = Hx + v$ , where  $H$  is a known matrix with mean value zero and covariance

$$P_{zz} = MH^T, \quad \bar{z} = H\bar{x}.$$

Problem 4, verify Equations (12.2.7) and the results of Problem 3. Verify the result!

Show that the gaussian random vectors are independent; i.e., we have  $E\{v_i v_j\} = 0$

number of theoretical relationships between state variables  $x$  is less than the number of measurements. For example, we have

$$Az = Hx + Av,$$

$q < p$ ,  $H$  is a  $(q \times p)$ -matrix,  $(vv^T) = R$  is a  $(p \times p)$ -matrix.

procedure of this section applies to the case where  $ARA^T$ .

ual problem of least squares estimation.

$$J = \frac{1}{2} \|z - Hx\|^2.$$

the fit  $e = z - H\hat{x}$  is orthogonal to the measurement error  $e = 0$ .

suppose that initial estimate  $\bar{x}_0 = 0$ ,  $\bar{\omega} = 2\pi/T$ ,  $\bar{\alpha}_0 = 0$ . Then, from Equations (12.2.7) and (12.2.8), the estimate of  $x_1$  is given by  $\hat{x}_1$ , where

$$\hat{x}_1 = \bar{x}_1 + P_1 H_1^T R_1^{-1} (z_1 - H_1 \bar{x}_1), \tag{12.3.6}$$

7 deg at  $t = 1.357$  sec.

$$E(\alpha - \bar{\alpha})^2 = 10^{-2} (\text{deg})^2, \quad E(a - \bar{a})^2 = 10^{-4} (\text{miles})^2, \quad E(e - \bar{e})^2 = 10^{-4},$$

$$E(T_0 - \bar{T}_0)^2 = 10^2 (\text{sec})^2, \quad E(\omega - \bar{\omega})^2 = 10^{-10} (\text{sec})^{-2},$$

$$E(\alpha_0 - \bar{\alpha}_0)^2 = 10^{-2} (\text{deg})^2,$$

all covariances are zero. [HINT: Use Equation (a) of Problem 2, and Equation (12.2.8), to find  $P$ .]

**Optimal filtering for single-stage linear transitions**

Consider a system that makes a discrete transition from state 0 to state 1 according to the linear relation

$$x_1 = \Phi_0 x_0 + \Gamma_0 w_0, \tag{12.3.1}$$

where  $\Phi_0$  is a known  $(n \times n)$  transition matrix,  $\Gamma_0$  is a known  $(n \times r)$ -matrix.

$$E(w_0) = \bar{w}_0, \quad E(w_0 - \bar{w}_0)(w_0 - \bar{w}_0)^T = Q_0. \tag{12.3.2}$$

where  $w_0$  is thus a random vector, with mean  $\bar{w}_0$  and covariance  $Q_0$ . The state  $x_0$  is also a random vector, with mean  $\bar{x}_0$  and covariance  $P_0$ ; that is, we have

$$E(x_0) = \bar{x}_0, \quad E(\hat{x}_0 - x_0)(\hat{x}_0 - x_0)^T = P_0. \tag{12.3.3}$$

Furthermore,  $x_0$  and  $w_0$  are independent. From this information, it follows that  $x_1$  is also a random vector, and from Section 11.2, Equation (11.2.7) through (11.2.9), it has a mean value  $\bar{x}_1$  and a covariance matrix  $P_1$ , given by

$$\bar{x}_1 = \Phi_0 \bar{x}_0 + \Gamma_0 \bar{w}_0, \tag{12.3.4}$$

$$M_1 = \Phi_0 P_0 \Phi_0^T + \Gamma_0 Q_0 \Gamma_0^T. \tag{12.3.5}$$

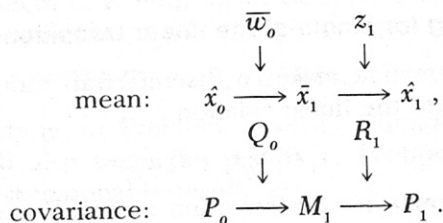
Since  $Q_0 \Gamma_0^T$  by definition (12.3.2) is a nonnegative matrix, it follows from (12.3.5) that, on the average, the effect of the uncertainty in  $w_0$  in a transition of the type (12.3.1) is to *increase the uncertainty* in our knowledge of the state  $x_1$ . † This is to be contrasted with the result (12.2.8), where it was shown that measurements, on the average, *decrease the uncertainty* in our knowledge of the state. †† We note that we make measurements, as in Section 12.2, *after* the transition to state 1. Then, from Equations (12.2.7) and (12.2.8), the estimate of  $x_1$  is given by  $\hat{x}_1$ , where

$$\hat{x}_1 = \bar{x}_1 + P_1 H_1^T R_1^{-1} (z_1 - H_1 \bar{x}_1), \tag{12.3.6}$$

precisely, the uncertainty is increased or left unchanged. The increase in  $P$  is dependent on the term  $\Phi_0$ . If  $\Phi_0$  is a rotation matrix, then, precisely, the uncertainty is decreased or left unchanged.

$$P_1 = (M_1^{-1} + H_1^T R^{-1} H_1)^{-1} = M_1 - M_1 H_1^T (H_1 M_1 H_1^T + R_1)^{-1} H_1 M_1 \quad (12.37)$$

Here  $\bar{x}_1$  and  $M_1$  are as given in (12.3.4) and (12.3.5). Note that  $\bar{x}_1$  is the estimate of  $x_1$  before measurement, whereas  $\hat{x}_1$  is the estimate after measurement. Similarly,  $M_1$  is the error covariance matrix before measurement and  $P_1$  is the error covariance matrix after measurement. Symbolically, we can describe this process as follows:



### 12.4 Optimal filtering and prediction for linear multistage processes

Consider the linear, stochastic, multistage process described by

$$x_{i+1} = \Phi_i x_i + \Gamma_i w_i, \quad i = 0, \dots, N-1, \quad (12.4.1)$$

where

$$E(x_0) = \bar{x}_0, \quad (12.4.2)$$

$$E(w_i) = \bar{w}_i, \quad (12.4.3)$$

$$E(x_0 - \bar{x}_0)(x_0 - \bar{x}_0)^T = M_0, \quad (12.4.4)$$

$$E(w_i - \bar{w}_i)(w_j - \bar{w}_j)^T = Q_i \delta_{ij}, \quad (12.4.5)$$

$$E(w_i - \bar{w}_i)(x_0 - \bar{x}_0)^T = 0. \quad (12.4.6)$$

Measurements  $z_i$  are made while the system is in stage  $i$ , and are linearly related to the state  $x_i$  by

$$z_i = H_i x_i + v_i, \quad i = 0, \dots, N, \quad (12.4.7)$$

where

$$E(v_i) = 0, \quad (12.4.8)$$

$$E(v_i v_j^T) = R_i \delta_{ij} \quad (12.4.9)$$

$$E(w_i - \bar{w}_i) v_j^T = 0, \dagger \quad \text{and} \quad E(x_0 - \bar{x}_0) v_i^T = 0. \quad (12.4.10)$$

It is reasonable to expect (see the derivation in Section 12.7 and Chapter 13) that the weighted-least-square or maximum-likelihood estimate of the state  $x_k$ , using only the measurements  $z_0, \dots, z_k$ ,

†The case in which  $w_i$  and  $v_j$  are correlated is considered in Chapter 13.

is given by the sequence of the previous section

$$\hat{x}_i = \bar{x}_i + K_i(z_i -$$

where

$$P_i = (M_i^{-1} + H_i^T R$$

This is the Kalman filter (Kalman, 1960). Note that the ratio between the measurements  $R_i$  and the error covariance matrix  $M_i$  is important. Note that:

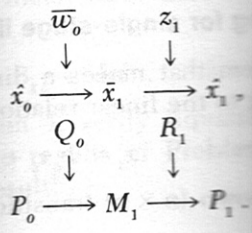
(a) The propagation of the error covariance matrix  $M_i$  is given by Equations (12.4.1) and (12.4.2). Thus,  $M_i$  is stored if the measurements  $z_i$  are given.

(b) The computation of the Kalman gain  $K_i$  and (12.4.12), in terms of the error covariance matrix  $M_i$ . Thus, the prediction of the state  $\hat{x}_{i+1}$  is available, say state  $\hat{x}_{i+1}$  that is,

$\hat{x}_{i+1} = \Phi_i \hat{x}_i + \Gamma_i \bar{w}_i + K_i(z_i - H_i \hat{x}_i)$  where  $\hat{x}_m$  is obtained from the previous step. In other words, the Kalman gain  $K_i$  is a function of  $w_i$ , namely,  $\bar{w}_i$ . However, with the filter, we consider  $R_i = \infty$  (12.4.15), (12.4.11).

†The term  $\bar{x}_{i+1}$  in (12.4.11) is  $E(x_{i+1})$ .

$M_1 H_1^T (H_1 M_1 H_1^T + R_1)^{-1} H_1 M_1$  (12.3.4) and (12.3.5). Note that  $\bar{w}_i$  is the measurement, whereas  $\bar{x}_i$  is the estimate.  $M_1$  is the error covariance matrix after measurement. The covariance matrix after measurement in this process as follows:



tion for linear multistage processes  
c, multistage process described by  
 $w_i, \quad i = 0, \dots, N-1,$

$$\begin{aligned}
 E(x_0) &= \bar{x}_0, \\
 E(w_i) &= \bar{w}_i, \\
 E(x_0 - \bar{x}_0)^T &= M_0, \\
 E(w_j - \bar{w}_j)^T &= Q_i \delta_{ij}, \\
 E(\bar{w}_i)(x_0 - \bar{x}_0)^T &= 0.
 \end{aligned}$$

while the system is in stage  $i$  by  
 $i, \quad i = 0, \dots, N,$

$$\begin{aligned}
 E(v_i) &= 0, \\
 E(v_i v_j^T) &= R_i \delta_{ij}
 \end{aligned}$$

† and  $E(x_0 - \bar{x}_0) v_i^T = 0$   
(see the derivation in Section 12.3)  
ted-least-square or maximum likelihood  
ing only the measurement

correlated is considered in Chapter 12

given by the sequential use of the single-stage estimation procedure of the previous section:

$$\hat{x}_i = \bar{x}_i + K_i(z_i - H_i \bar{x}_i), \quad (i = 0, \dots, k, \text{ where } k \leq N). \quad (12.4.11)$$

$$\bar{x}_{i+1} = \Phi_i \bar{x}_i + \Gamma_i \bar{w}_i, \quad \bar{x}_0 \text{ given}, \quad (12.4.12) \dagger$$

$$K_i = P_i H_i^T R_i^{-1}, \quad (12.4.13)$$

$$P_i = (M_i^{-1} + H_i^T R_i^{-1} H_i)^{-1} = M_i - M_i H_i^T (H_i M_i H_i^T + R_i)^{-1} H_i M_i, \quad (12.4.14)$$

$$M_{i+1} = \Phi_i P_i \Phi_i^T + \Gamma_i Q_i \Gamma_i^T. \quad (12.4.15)$$

This is the *Kalman filter* for linear multistage processes (see Kalman, 1960). Note that the filter (12.4.11) and (12.4.12) is a model of the system (12.4.1), with a correction term proportional to the difference between the actual measurement  $z_i$  and the predicted measurement  $H_i \bar{x}_i$ . The proportionality matrix  $K_i$  in (12.4.13) is essentially the ratio between uncertainty in the state  $P_i$  and the uncertainty in the measurements  $R_i$ ; the matrix  $H_i$  is simply the state-to-measurement estimation matrix of (12.4.7).

The propagation of the covariance of the error of the estimate, Equations (12.4.14) and (12.4.15), is *independent* of the measurements  $z_i$ . Thus, the covariance matrix can be computed *beforehand* if the parameters of the system and the observation equations are given.

The computation of the updated estimate, Equations (12.4.11) and (12.4.12), involves only the *current* measurement and error covariance. Thus, it can easily be carried out in real time.

*Prediction* of the state beyond the stage where measurements are available, say state  $m$ , can be done only by repeated use of (12.4.12);

$$\hat{x}_{i+1} = \bar{x}_{i+1} = \Phi_i \hat{x}_i + \Gamma_i \bar{w}_i; \quad i = m, m+1, \dots, \quad (12.4.16)$$

where  $\bar{x}_m$  is obtained from the filter (12.4.11) through (12.4.15). In other words, the best prediction we can make uses the expected value of  $\bar{w}_i$ , namely,  $\bar{w}_i$ , in the transition relations (12.4.1), starting, however, with the filtering estimate of  $\hat{x}_m$ . Another way of seeing this is to let  $R_i = \infty$  for  $i = m, m+1, \dots$ . In this case, (12.4.14) and (12.4.15), (12.4.11) and (12.4.12) reduce to

$$P_{i+1} = \Phi_i P_i \Phi_i^T + \Gamma_i Q_i \Gamma_i^T, \quad (12.4.17)$$

where  $\bar{x}_{i+1}$  in (12.4.12) is, of course, to be understood as  $E(x_{i+1}/z_i, \dots, z_i)$  and not