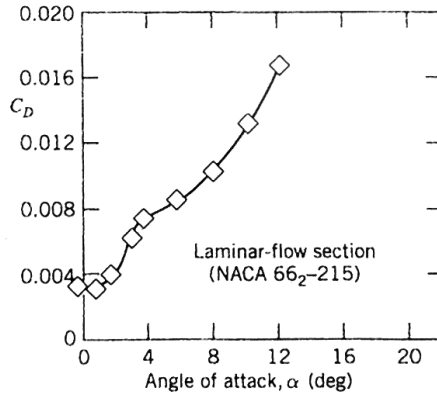
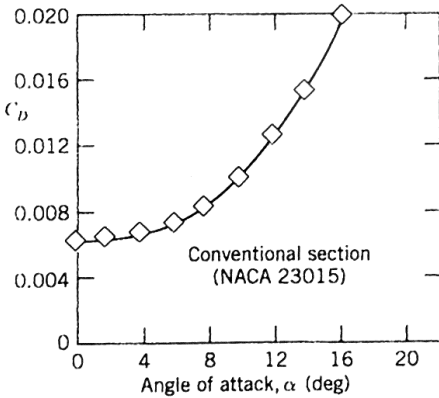


(a) Lift coefficient vs. angle of attack



(b) Drag coefficient vs. angle of attack

Lift and drag coefficients versus angle of attack for two airfoil sections of 15 percent thickness ratio at $Re_c = 9 \times 10^6$. (Data from Abbott and von Doenhoff [21].)